A REPORT

ON

Development of the flight model of an inertial asset control system for ARAMIS satellite

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ABSTRACT

The present report discusses the work carried out on the project "Development of the flight model of an inertial asset control system for ARAMIS satellite". ARAMIS is the second satellite being developed at Politecnico di Torino after developing a satellite called PiCPoT (Piccolo Cubo del Politecnico di Torino) in July 2006, which had as its main purposes the test on the operation of commercial components (COTS) in space and the acquisition of images and spatial parameters. The present Satellite ARAMIS will be one step ahead of PiCPoT and the central concept kept in mind is to introduce the concept of modularity of the satellite which will significantly reduce costs of design and development of the satellites. Defining various standard modules which are compatible with each other and then assembling these modules in the quantities and manner required by the mission.

Among the various modules in the ARAMIS Satellite one is the Attitude and Orbit Control Systems (**AOCS**) subsystem which is one of the most important subsystems of a satellite. Its main function is to control the attitude and the orbit of the satellite. It is often the most complex of the satellite subsystems. Its function is to control the attitude and the orbit of the satellite. This is a critical function because without attitude or orbit control the satellite will gradually – sometimes in a matter of seconds, sometimes after several days – become unable to fulfil its mission objectives.

AOCS' chief task is to periodically collect measurements from sensors and convert them into commands for actuators.

There are two major Attitude Actuators Subsystem(1B21) in the ARAMIS which are:

- 1) Magnetic Torque Actuator (1B211)
- 2) Reaction Wheel Actuator (1B212)

Also there are two major Attitude Sensor Subsystem(1B22) which are:

- 1) Magnetometer Sensor (1B221)
- 2) Gyroscopic Sensor (1B222)

This project aimed towards development of all the necessary Software and Hardware for the 'Reaction Wheel Actuator' and 'Gyroscopic Sensor'. While all the necessary software and hardware for other two module namely 'Magnetic Torque Actuator' and 'Magnetometer Sensor' are been developed in an another project.

The basic principal which governs the control of orbit and rotation of the satellite is the law of conservation of angular momentum. A reaction wheel placed inside the satellite is rotated at a controlled speed for a calculated duration. This in turns rotates the satellite in opposite direction in order the conserve the total Angular momentum of the satellite.

The code has been written for the microcontroller (TI MSP430) which reads the gyroscopic sensor, controls the reaction wheel and communicates with the central processor of the satellite. The verification has been done by downloading the code onto the circuit with the microcontroller TI MSP430F2274. All the modules necessary for the communication with the central processor have been written and tested. The modelling for the system has been done in UML through Visual Paradigm and the compilation, simulation and assembly code generation along with downloading onto the chip has been done by IAR Embedded Workbench for MSP430.

All the newly developed class-diagrams, Sequence diagrams and Use-case implementations have been integrated into the overall ARAMIS satellite code and model.

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1. INTRODUCTION

The Politecnico di Torino has developed a satellite called PiCPoT (Piccolo Cubo del Politecnico di Torino) in 2006, which had as its main purposes the test on the operation of commercial components (COTS) in space and the acquisition of images and spatial parameters of . In addition to these goals of scientific-technical, the establishment of this satellite has also had the advantage of involving many departments of the Politecnico di Torino in the design of the modules that make up the satellite involving teachers, researchers, students and doctoral candidates. The satellite PiCPoT has a cubic shape of side about 12 cm. In five of the six faces of PiCPoT there is a solar panels that convert solar energy into electrical energy used by electronics on board, while the sixth face of the cube have two antennas used for communication with the ground station and the cameras. The creation of this satellite was completed in July 2006 with the launch of the satellite through the Russian Dnepr carrier. Due to a problem of the hydraulic pitcher has not been possible to verify the behaviour of the satellite in orbit. With the experience gained during the design of the satellite PiCPoT have created the basis for a second project, the satellite Aramis. The project had the main PiCPoT "defect" to be developed ad-hoc for the mission, so we had a usability project very low which made the high cost of design. By ARAMIS satellite we want to introduce the concept of modularity of the satellite to remedy the defects in PiCPoT. Defining the standard modules are compatible with each other, the modules can be assembled in the quantities and manner required by the mission. This approach reduces costs of design and development as the modularity of design allows the use of modules in different missions with different architectures thereby spreading the cost of the project.

ARAMIS project contains various modules which have been divided into various categories. Figure 1 shows the various modules and sub-modules defined in ARAMIS.

This report deals with the design, implementation and testing of the 'Attitude and Orbit Subsystem' (1B2 in figure 1) of the satellite ARAMIS.



Figure 1: ARAMIS Modules

The detailed description of the 'Attitude and Orbit Subsystem' module is shown in Figure 2. Among the various other things, the present work focuses on design and implementation of the submodules 'Reaction Wheel Actuator', 'Gyroscopic Sensor' and 'Centralized Attitude Controller'. The position and link between all these sub-modules can be seen in the figure 2.



Figure 2: Attitude and Orbit Subsystem

2. DESCRIPTION OF THE DEVICES

Various devices has been used in designing and implementing the required system. The major devices among them are:

- 1) Microcontroller
- 2) Motor
- 3) Brushless Motor Driver
- 4) Gyroscope

2.1) Microcontroller

The controller must have the necessary features to manage the two drivers, sensors, A / D conversions and communication with the outside (in UML called AOCS).

The controller must have the following characteristics:

- 6-pin to operate the brushless motor driver
- 5 pin to operate the solenoid driver.
- 1 timer for timed feeding of the solenoid
- 1 timer for timed circuit set / reset of the magnetometer
- 2 pin to force the signals to set and reset the magnetometer
- 1 A / D converter.
- 2-channel (pin) for ADC for the signals from the gyroscope
- 2-channel (pin) for ADC for the signals from the magnetometer
- 1 channel (pin) for ADC for the signal from the solenoid driver.
- 1 UART communication interface used for communication and testing.
- low consumption.

So the minimum features that the controller must have are:

- at least 2 timers.
- an ADC with at least 5 channels.
- 12-pin general purpose.
- A UART interface.
- low power consumption.

The controller is chosen **MSP430F2274** (family MSP430x2xx) of Texas Instruments, which has the following characteristics:

- Low voltage (1.8 V to 3.6 V).
- Low Power (Active mode: 270µA at 1MHz, 2.2 V Standby mode: 0.7 µA).

CPU • 16-bit RISC.

- Configurable Clock up to 16MHz internally.
- 2 timers (Timer_A, Timer_B) to 16 bits.
- Universal Serial Communication Interface (UART, SPI, I2C).
- A / D 10-bit, 200-Ksps, internal voltage reference.
- Package 38-pin TSSOP.

The pinout of this microcontroller is in the Figure 3 below.



Figure 3: Pinout of the Microcontroller MSP430F2274



The schematic of this structure is shown in the figure 4:

Figure 4: Schematic of the Microcontroller MSP430F2274

As can be seen in the schematic diagram, the microcontroller is powered with a voltage of 3.3 V via pins 2 and 16. On pin 5 and 6 has been connected to a quartz X1 (SE2418CT) with resonance frequency

of 32.768 KHz with two external capacitors (C20 and C21) of the value of 12pF, as recommended by the datasheet of quartz.

Connectors J11 and J14 connectors are used for programming (microcontroller) via 4-wire JTAG and JTAG 2-Wire (Spy Bi-Wire).

The J12 connector is the connector used for serial communication.

The A / D converter will be operated with a clock, using the internal oscillator to the microcontroller.

2.2) Motor

In order to rotate the wheel of inertia a reaction wheel or motor is required. The motor chosen to be used in ARAMIS is brushless motor of the EC 32 flat Maxon Motor. This motor fits best to our requirements of high reliability, small weight, high torque and easily integrable into the satellite.

Various details of the motor is in the next diagram:

EC 32 Flat motor Ø32 mm, brushless, 6 Watt

flat motor maxon





Order Number





Details on page 149

M 1:2

PIN 1

12,1 +4,5



01140

Specifications **Operating Range** Axial proload > 5 N n [rpm] 160001 ٠ Axial preioad Max, ball bearing loads axial (dynamic) radial (7.5 mm from flange) Force for press fits (static) (static, shaft supported) 2.8 N 5.5 N 50 N 14000 12000 1000 N 10000 ٠ Ambient temperature range -40 ... +100°C 8000 ٠ Max permissible winding temperature +125°C 6000 ٠ Weight of motor 32 g 4000 Version with and without Hall sensors ٠ 2000 ٠ 8 pole permanent magnet ٠ 3 phased coil stator with 2 pole shoes each 2 6 8 0.4 0.6 0.8 • Values listed in the table are nominal. ٠ Connection with Hall sensors sensorless With Hall sensor 3.5...20 VDC Hall sensor 3 Hall sensor 2 GND Motor winding 3 Motor winding 1 Order number Pin 1 Pin 2 Pin 3 Pin 4 Pin 5 Pin 6 Pin 7 Pin 8 Adact Motor winding 1 Motor winding 2 Motor winding 3 , neutral point 0.4 0.6 0.2 0.8 Order number Order number Adapter see p. 275 Connector 220300 Article number 220310 Article number ٠ AMP MOLEX MOLEX 1-487951-1 52207-1190 52089-1110 487951-4 52207-0490 52089-0410 Pin for design with Hall sensors: FPC, 11 pole, pitch 1.0 mm, top contact style For wining diagram for Hall sensors, see p. 26 184 maxon EC motor



Figure 5: Motor Datasheet



The schematic of the connections of the overall circuit made in order to control the reaction wheel is as shown in the figure:

Figure 6: Schematic of overall circuit

2.3 Brushless Motor Driver

It is necessary to control the brushless motor so that it does rotate the wheel of inertia at a speed imposed by the desired angular momentum. The control is imposed by the means of a Motor Driver which can communicate with the processor and accordingly gives signals to the motor. To drive the motor you chose to use the driver for brushless motors A8904 of Allegro Microsystems. <u>Its main features are :</u>

- Typical voltage power control logic at 5V.
- Supply voltage of the load (motor) from 4 to 14V.
- Integrated three-phase bridge.
- Ability to provide output currents up to 1.4 A (3A peak)
- Programming serial speed and direction of rotation
- · Serial programming of the maximum current absorbed by the load
- Programming of the serial number of poles of motor
- Integrated speed control based on back-EMF detection
- Package 24-pin SOIC.

The pin driver is shown in Figure 7.



Figure 7: Pin Driver A8904

Serial Programming of the Brushless Motor Driver

The commands are sent from the microcontroller through the serial 29 bit.

When CHIP SELECT is low, the data is placed on the serial port on the rising edge of the clock with the MSB (D28) first.

At the end of the programming of the driver CHP SELECT returns to the logical value high. The timing to be observed are given in the following figure 8.

Serial Port Timing Conditions



Α.	Minimum CHIP SELECT setup time before CLOCK rising edge	100 ns
В.	Minimum CHIP SELECT hold time after CLOCK rising edge	150 ns
C.	Minimum DATA setup time before CLOCK rising edge	150 ns
D.	Minimum DATA hold time after CLOCK rising edge	150 ns
Ε.	Minimum CLOCK low time before CHIP SELECT	. 50 ns
F.	Maximum CLOCK frequency	3.3 MHz
G.	Minimum CHIP SELECT high time	500 ns

Figure 8: Timing for programming driver A8904

Through the 29-bit (D0. .. D28) programming, you can set the maximum current that can absorb the engine, the number of poles of the motor, the direction and speed of rotation and more. The programming of the desired speed is achieved using 14-bit (D5. .. D18), these bits are used to set the total count expresses the number of oscillations must count the speed control to the desired speed, i.e. :

 $total \ count = \frac{(60 \cdot f_{OSC})}{desidered \ motor \ speed \ (rpm)}$

To set this speed should be set to logical high value bits D14, D13, D11, D10, D9, D8, D7, D6 and D5.

Bit number	Count number
D5	16
D6	32
D7	64
D8	128
D9	256
D10	512
D11	1,024
D12	2,048
D13	4,096
D14	8,192
D15	16,384
D16	32,768
D17	65,536
D18	131,072

Figure 9: bit values associated with the programming



The schematic of the Brushless Motor Driver is as follows:

Figure 10: Schematic of the Brushless Motor Driver

2.4 Gyroscope

Gyroscope measure the temperature of the system and the angular rate of the satellite.

The gyro sensor must be able to measure the angular velocity around the axis perpendicular to the tile of PowerSupply. Maintaining the same system of reference of the magnetometer, i.e., assuming that the solenoid and then printed lies at x, y, the gyroscope must acquire the angular velocity around the z axis as shown in following figure 11.



Figure 11: Axis measuring gyro sensor

This sensor must be capable of measuring speed of rotation of the satellite equal to ± 2 rpm ($\pm 12^{\circ}$ /s).

2.4.1) Gyro sensor

The characteristics that the required sensor must have are:

- Measuring z axis (yaw-rate).
- Low consumption.
- Measuring range greater than or equal to \pm 12 ° / s.

The sensor chosen is the ADXRS401 Analog Device.

Its main features are:

• Measurement of the angular velocity around the z axis, i.e. the axis perpendicular to the surface mounting of the sensor.

- Supply Voltage 5V typical.
- Measuring range \pm 75 ° / s.
- Output voltage at 0 ° / s typical 2.5 V.
- Typical Sensitivity 15mV / ° / s.
- Integrated temperature sensor.
- Package 7mm x 7mm x 3mm 32-pin BGA surface mount.
- Consumption of up to about 8MA.

The pinout of this sensor is shown in following figure which represents the component side seen from the bottom:



Figure 12: Gyroscopic sensor seen from the bottom

The 32-pin BGA component are reported in pairs in 16-pin logic is 12-pin physical interior is a replica of the external pin.

This solution probably has been adopted to make the PCB mounting more stable, without increasing the footprint of the component.

Referring to the pinout above, the gyro sensor output will be::

$$V_{ADXRS401_{RATEOUT}} = V_0 + S \cdot K_s$$

where:

- K_{S} represents the sensitivity of the gyro sensor in mV / ° / s.
- *S* is the angular velocity in $^{\circ}$ / s, which varies from -12 $^{\circ}$ / s at +12 $^{\circ}$ / s.
- V_0 is the tension $V_{ADXRS401_{RATEOUT}}$ with S = 0.

Built in gyro sensor is a temperature sensor, whose output voltage is:

$$V_{ADXRS401_{TEMP}} = K_T \cdot (T - 25) + 2.5V$$

where:

- K_T represents the sensitivity of the temperature sensor in mV / ° C.
- T is the temperature expressed in ° C ranging from -40 ° C to +85 ° C. 85 °C.

2.4.2) Conditioning circuit

The schematic circuit packaging is the figure below.



Figure 13: Gyroscopic schematic

The output voltages of the sensor must be scaled so that it can be acquired from the converter analog / digital controller.

It is therefore necessary to constrain the two voltages so that they remain in a range between 0 and 2.5 V.

The 0V represent the minimum output of the gyroscope and the 2.5V represents the maximum output of the gyroscope.

The above is an important condition as the code in the Gyroscope Class has been written assuming the above condition is fulfilled.

3. <u>SYSTEM SPECIFICATIONS</u>

Use Case Diagram

ATTITUDE CONTROL SUBSYTEM (1B232 CENTRALIZED ATTITUDE CONTROLLER)

The following Use-Case diagram shows the use-cases which has to be implemented in the 'Centralized Attitude Controller' subsystem.

Various Use-Cases has been categorized in terms of the action they perform, i.e. send a command to the module, get the satellite housekeeping data from the module containing the information, send a command to read / write some data, set the configuration of the module etc.



Figure 14: Use Case Diagram

The detail/purpose of each Use-case is as follows :

Name	Documentation
1B232 Centralized Attitude Controller	The object running on the On-Board Computer which controls the attitude of the whole satellite.
Module Commands	The CPU actor sends a data-less command to the System .
	The CPU can use up to 8 different Module Command commands, to issue as many commands to the system.
	It uses the SPI Protocol - Command Only use case, by issuing the CMD_COMMAND_x command, where \mathbf{x} (07) identifies the command type; \mathbf{x} does not identify the sequence in which messages are written, but the Designer -defined type (therefore the effect of the command).
	The Designer can use as many message types he wants.
	This use case is optional, therefore the Configurator shall #define an identifier whose name is contained in the tagged value define .
EnableWheel	Enables reaction wheel. Any further action of reaction wheel will be carried on.
	Uses Module_Commands use case of 1B45 package, with the CMD_COMMAND_0 command.
DisableWheel	Disables reaction wheel. Any further action of reaction wheel will be disregarded. If the reaction wheel is in process of rotation while the DisableWheel command is given, the wheel is stopped immediately.
	Uses Module_Commands use case of 1B45 package, with the CMD_COMMAND_1 command.
deactivateGyroscope	Disables gyroscope (removing power supply).
	Uses Module_Commands use case of 1B45 package, with the CMD_COMMAND_2 command.
activateGyroscope	Enables gyroscope (giving power supply and sampling its output).
	Uses Module_Commands use case of 1B45 package, with the CMD_COMMAND_3 command.

Get Module Housekeeping	Returns last measured housekeeping data (see use case Housekeeping for details).
	It uses the SPI protocol - Read Data use case by issuing the CMD_GET_HOUSEKEEPING command.
	This use case is optional, therefore the Configurator shall #define an identifier whose name is contained in the tagged value define .
getSpin	It measures and returns the rotational speed around z axis with a signed 16-bits integer; unit is 0.0001 rad/s.
	It provides housekeeping data (spin) to Get Module Housekeeping use case of 1B45 package, in the Housekeeping_commons[0] variable.
getTemperature	It measures and returns the temperature with a signed 16-bits integer; unit is 0.01°C.
	It provides housekeeping data (temperature) to Get Module Housekeeping use case of 1B45 package, in the Housekeeping_commons[1] variable.
Read Module Data	The CPU actor reads up to 256B of Designer -defined data from the System .
	The CPU can use up to 8 different Read Module Data commands, to read messages from as many different subsystems. It uses the SPI Protocol - Read Data use case, by issuing the CMD_READ_DATA_x command, where x (07) identifies the message type; x does not identify the sequence in which messages are read, but the Designer -defined type (therefore the source subsystem).
	The Designer can use as many message types he wants.
	This use case is optional, therefore the Configurator shall #define an identifier whose name is contained in the tagged value define .
isRotating	Checks whether the reaction wheel is in process of rotation with respect to satellite or is stopped.
	Returns 1 if the motor is running and returns 0 if the motor is stopped.
	Uses Read Module Data use case of 1B45 package, with the CMD_READ_DATA_0 command which returns a 1-byte message containing the status of the reaction wheel.

Write Module Data	The CPU actor sends up to 256B of Designer -defined data to the System .
	The CPU can use up to 8 different Write Module Data commands, to issue messages to as many different subsystems.
	It uses the SPI Protocol - Write Data use case, by issuing the CMD_WRITE_DATA_x command, where $x (07)$ identifies the message type; x does not identify the sequence in which messages are written, but the Designer -defined type (therefore the destination subsystem).
	The Designer can use as many message types he wants.
	This use case is optional, therefore the Configurator shall #define an identifier whose name is contained in the tagged value define .
actuateReactionWheel	It generates a user-defined angular momentum by means of the reaction wheel. The momentum is generated around z axis. Positive momentum will push x axis towards y axis
	If angular momentum is zero, the wheel is stopped.
	If momentum is higher than max allowed, wheel will rotate at max speed.
	If momentum is lower than min allowed, wheel will rotate at min speed.
	Angular momentum is signed 16-bits integer in units of 10 gcm2/s.
	Uses Write Module Data use case of 1B45 package, with the CMD_WRITE_DATA_0 command with a 2-byte message containing the angular momentum with MSB first/last (TBD).
rotateReactionWheel	rotates reaction wheel by a user-defined number of turns.
	Angle is a signed 16-bits integer; unit is 0.1kgcm2rad
	Uses Write Module Data use case of 1B45 pacvkage, with the CMD_WRITE_DATA_1 command with a 2-byte message containing the speed (in rpm) at which the reaction wheel has to be rotated, with MSB first/last (TBD).
onflightCalibration	Allows to change value of wheel inertia, offset and sensitivity of gyroscope, using three signed 16-bits integers. Units are: TBD, TBD, TBD.
	Uses Set Module Configuration use case of 1B45 package, with:
	• gyroscope spingain in field gysoscope_gain_s,
	• gyroscope spin offsetin field gysoscope_offset_s,

	 gyroscope temperature gain in field gysoscope_gain_T, gyroscope temperature offsetin field gysoscope_offset_T, Moment of Inertia of the reaction wheel in Inertia_motor.
Set Module Configuration	Changes some TBD Designer -defined configuration parameter (if any).
	The Designer shall define all configurable parameters (if any) in the type t_Configuration .
	This use case is different from the Configure Module use case, as the latter allows a compile-time configuration, while this use case allows run-time configuration changes (if foreseen).
	This use case is optional, therefore the Configurator shall #define an identifier whose name is contained in the tagged value define .

4. SYSTEM DESIGN

Sequence Diagram

MOTORDRIVER - ROTATE





5. IMPLEMENTATION

1B21 Class Diagram :

The following is the class diagram developed for implementing the system. The details for each specific class can be found in the next section.



Figure 16: Class Diagram

5.1 Description of Classes and Packages

The description of various classes and packages used in the implementation of the system is as follows:

Name	Documentation
Common	
MotorDriver	Controls the motor by setting the desired speed to the motor. It generates and sends necessary signals to the motor driver 8904 which in turns control the motor.
Bk1B45_Slave	This package contains all specifications of the Peripheral (slave) side of the Bk1B45 Subsystem Serial Data Bus. In these diagrams, the System is the Peripheral side.
	It comprises two Use Case diagrams:
	• Housekeeping Use Case Diagram, which incorporates all use cases related to reading from the slave its housekeeping data and the related history and statistics, system status; to issue commands to the system like wake-up, standby and reset, and to read/write application- specific (Designer-defined) data.
	• Supply, Enable, Configuration Use Case Diagram, which incorporates use cases related to power supply, static configuration and testing of the system.
	The document only describes commonly-used functions of a Peripheral (slave), and the Designer may add as many functions as he requires. Yet any added function should comply as much as possible to the basic protocol described herein.
	Furthermore, not all functions described in these use case diagrams need be implemented. Most use cases are optional. If used, they shall be implemented as specified. If not used, they can be disabled by removing appropriate attributes from the relevant classes, as indicated in the Configure Module use case.
Buffers	This class implements a set of fixed-length buffers, to be used for Write Module Data and Read Module Data use cases, respectively. Up to MAXBUFFERS buffers of 256 bytes each can be allocated (provided that the processor has enough memory).
	Each buffer can be used for either Write or Read operation, although the class guarantees that at least one buffer is allocated for both Write and Read use case.
	In particular, for either Write (if command =

	CMD_WRITE_DATA_x) or Read (for command = CMD_READ_DATA_x):
	• throws away any buffer of that type still being written but not yet complete (namely, for which the ready() operation has not yet been called, if any)
	• finds the first available buffer of that type
	• locks it and declares it being written; the user shall then fill the buffer and call the ready() operation when finished. After that, the buffer is queued
	• returns a pointer to it
	Returns null if no buffer of the chosen type is available or command is not supported.
t_Commands	Lists all available command codes and the corresponding value. There is one command for each use case,
	Removing a command from this class removes the corresponding use case.
	Adding additional commands requires adding the appropriate code to properly interpret and execute the command.
Test	for testing the motordriver class. It sends various commands to the motordriver class.
t_sensor	The type to be used to store sensor data
CommandInterpreter	Interprets various commands.
Housekeeping	Does the housekeeping related activities, constantly monitors housekeeping parameters.
Gyroscope	Measures the spin and the temperature.
ADC	Converts analog signals into Digital signals capable of being read/write by processor.
t_Configuration	Defines various on-flight configurable parameters.
TimerA	Timer for measuring and calculating time.

5.2 Details of major classes

Some of the classes like MotorDriver have been newly developed, some other classes like Housekeeping, CommandInterpreter etc. have been derived from 1B45 project and redefined or extended, while few classes like Timer, ADC & Buffer have been simply used from 1B45 project.

The classes which have been newly developed are:

- 1) MotorDriver
- 2) Gyroscope

5.2.1 MotorDriver

Controls the motor by setting the desired speed to the motor. It generates and sends necessary signals to the motor driver 8904 which in turns control the motor.

MotorDriver Class consists of :

- 1) Attributes
- 2) Operations

MotorDriver ->Attributes

<u>private PORT_RESET : port_address = & P3OUT</u>		
Documentation	Selects the port on which the reset signal will be sent to the motor driver 8904.	
private BIT_RESE	$\underline{T: byte = BIT3}$	
Documentation	Select the pin for the BIT_RESET function on the PORT specified.	
private PORT CH	<u>IP_SELECT : port_address = &P3OUT</u>	
Documentation	Selects the port on which the chip select signal will be sent to the motor driver 8904.	
private BIT_CHIP_SELECT : byte = BIT6		
Documentation	Select the pin for the CHIP_SELECT function on the PORT specified.	
<u>private PORT_CLOCK : port_address = & P1OUT</u>		
Documentation	Selects the port on which the clock signal will be sent to the motor driver 8904.	
private BIT_CLOCK : byte = BIT5		
Documentation	Select the pin for the CLOCK function on the PORT specified.	
<u>private PORT_DATA_OUT : port_address = &P1OUT</u>		
Documentation	Selects the port on which the data out signal will be sent to the motor driver 8904.	
private BIT_DATA_OUT : byte = BIT6		
Documentation	Select the pin for the DATA_OUT function on the PORT specified.	

private OSC FREQ : ulong = 2000000		
Documentation	Frequency of Oscillator used in Hz.	
private POLES : us	short = 4	
Documentation	Number of poles of motor. Shall be either 4, 8, 12 or 16.	
private CURRPRO	OG: short = 0	
Documentation	Current limitation and transconductance gain:	
	0:1.2A; 500mA/V	
	1:0.6A;250mA/V	
	2:1A; 500mA/V	
	3:0.5A;250mA/V	
	4:0.6A; 500mA/V	
	5:0.3A;250mA/V	
	6:0.25A; 500mA/V	
	7:0.125A;250mA/V	
private CURRWA	<u> TCHDOG : ushort = 40</u>	
Documentation	Absolute value of charge current (in uA) for the watchdog timer. Shall be either 10, 20, 30 or 40.	
public motor_runn	ing_time : long = 0	
Documentation	this variable stores the time remaining in ms for which the motor has to be rotated.	
<u>private motor_enal</u>	<u>ble : byte</u>	
Documentation	it decides where the motor is enabled or disabled.	
	motor_enable==0 means that motor is disabled and any command given to motor will be disregarded.	
	motor_enable==1 means that motor is enabled and any command given to motor will be executed.	
private SPEED : short = 3000		
Documentation	speed in rpm at which of the motor should be rotated by the commands rotateReactionWheel() and actuateReactionWheel().	
	presently choosen as 3000	
	More efficient decision of the motor speed according to the running time can be taken at a later stage by the Designer.	
public BIT_MOTO	DR_ROTATING : byte const = BIT4	
Documentation	set the bit in t_Status:statusWord which corresponds to the state of the motor(rotating/stop).	

MotorDriver ->Operations

<pre>public init_motor()</pre>	
Code	*(PORT_CHIP_SELECT + (&P1DIR-&P1OUT)) = BIT_CHIP_SELECT;
	*(PORT_CLOCK + (&P1DIR-&P1OUT)) = BIT_CLOCK;
	*(PORT_DATA_OUT + (&P1DIR-&P1OUT)) = BIT_DATA_OUT;
	*(PORT_RESET + (&P1DIR-&P1OUT)) = BIT_RESET;
	clock(0); chip_select(1);
	reset();
	motor running time=0:
	status.statusWord = status.statusWord & (~BIT_MOTOR_ROTATING);
	//set the MOTOR_ROTATING status to stop config.Inertia_motor = 413.9;
Documentation	Initializes the signals necessary to run the motor.

public rotate(speed : short)				
Parameter	speed			
	Multiplicity	Unspecified		
	Documentation	the speed in rpm at which the motor should be rotated.		
	Туре	short		
	Direction	in		
Code	<pre>if(speed==0) status.statusWord = status.statusWord & (~BIT_MOTOR_ROTATING); else status.statusWord = status.statusWord BIT_MOTOR_ROTATING; generatepattern(speed);</pre>			
Documentation	Drives the motor to rotate clockwise (if speed is positive) or counterclockwise (if speed is negative) at angular speed equal to speed rpm. The routine terminates immediately, while the motor keeps on rotating until next			
	command.			

public actuateReactionWheel(moment : long) : void			
Parameter	moment		

	Multiplicity	Unspecified		
	Documentation	the moment which the motor should provide. it is used for calculating the time at which the motor has to be rotated at SPEED rpm.		
		motor_running_time =1000* moment / ((config.Inertia_motor + Inertia_Satellite) * SPEED * (2 * 3.14 / 60));		
	Туре	long		
	Direction	in		
Code	//computer the time in ms for which motor has to be rotated.			
	motor_running_time =1000* moment / (config.Inertia_motor * SPEED * (2 * 3.14 / 60));			
	rotate(SPEED);			
Documentation	turns the satellite by the specified moment using the rotation of motor.			
	the motor speed is set to SPEED rpm.			
	If doesn't check if the motor is already running or not, which has to be checked by the processor giving the command.			
	The routine terminates immediately, while the motor keeps on rotating for the required time.			

public rotateReactionWheel(turns : short)							
Parameter	turns						
	Multiplicity Unspecified						
	Documentation Number of turns the wheel should rotate.						
	Type short						
	Direction	in					
Code	<pre>motor_running_time = 1000 * turns / (SPEED *60); rotate(SPEED);</pre>						
Documentation	Rotates the reaction wheel by a user-defined number of turns specified in the parameter turn .						
	The routine terminates immediately, while the motor keeps on rotating for the required time.						

public stopRotation()			
Code	rotate(0);		
Documentation	Drives the motor to stop immediately.		

public generatepat	ttern(speed : short)	
Parameter	speed	
	Multiplicity	Unspecified
	Documentation	speed is the angular speed in rpm with which the motor should be rotated.
	Туре	short
	Direction	in
Code	unsigned long prog char stop, direction	=0, count=0; ;
	//init_motor();	
	<pre>if (speed == 0) { stop = 1; count =0; direction = 0; } else if (speed > 0) { count = 60*OS direction = 0; stop = 0; } else { count = 60*OS direction = 1; stop = 0; } }</pre>	{ SC_FREQ/speed; SC_FREQ/ (- speed);
	// compute the prog prog = (((count/16) 2 (motor_enable & prog = (CURRPRO ((long)CURRPRO switch (CURRWA case 10: break; case 20: prog = (lo case 30: prog = (lo case 40: prog = (lo	<pre>gramming word & 0x1FFF) << 5) ((long) direction) << 25 ((long) stop) << & 0x1); DG & 0x4) << 1 (CURRPROG & 0x2) << 3 G & (long)0x1) << 28; TCHDOG) { ng)0x2 << 26; break; ng)0x1 << 26; break; ng)0x3 << 26; break;</pre>

	<pre>} switch (POLES) { case 4: prog = (long)0x1 << 21; break; case 8: break; case 12: prog = (long)0x3 << 21; break; case 16: prog = (long)0x2 << 21; break; } clock(0); chip_select(0);</pre>
	sendpattern(prog); chip_select(1);
Documentation	This function generates a 29-bit pattern code and using sendpattern() send it to the 8904(motor driver) in order to program it to rotate the motor at the specified speed .

public sendpattern(pattern : long)				
Parameter	pattern			
	Multiplicity	Unspecified		
	Documentation	pattern is the 29-bit programming word which is sent to the motor driver(8904) in order to set various parameters of the motor.		
	Туре	long		
	Direction	inout		
Code	for(i = 28; i >= 0; i) send_bit(pattern & (((long) $0x1) \ll i$));			
Documentation	Send the 29 bit programming word pattern to 8904 motor driver. This 29-bit word controls various parameters of reaction wheel like: rotation speed, direction, watchdog current, brake, sleep mode, step mode etc.			

public send_bit(value : long)			
Parameter	value		
	Multiplicity	Unspecified	
	Туре	long	
	Direction	in	
Code	data_out(value); clock(1); clock(0);		
Documentation	Send a bit to the motor driver's data_in port.		

private data_out(value : long)							
Parameter	value						
	Multiplicity Unspecified						
	Documentation	valuewill be checked for zero or non-zero condition.					
	Туре	long					
	Direction	in					
Code	*PORT_DATA_OUT =(*PORT_DATA_OUT & ~ BIT_DATA_OUT) (value ? BIT_DATA_OUT: 0);						
Documentation	Send the data (29-bit programming word) serially to the motor driver (8904).						

private chip_select(value : byte)						
Parameter	value					
	Multiplicity	Unspecified				
	Documentation value will be checked for zero or non-zero condition.					
	Туре	byte				
	Direction	in				
Code	*PORT_CHIP_SELECT = (*PORT_CHIP_SELECT & ~ BIT_CHIP_SELECT) (value ? BIT_CHIP_SELECT: 0) ;					
Documentation	Controls the chip select signal on the motor driver (8904).					
	Sets chip select to 1 when value $<>0$; else sets chip select to 0.					

nrivate d	lock	(val	ne	•	hvte)
private	JUCK	V al	uc	•	<u>Ujtej</u>

Parameter	value	
	Multiplicity	Unspecified
	Documentation	valuewill be checked for zero or non-zero condition.
	Туре	byte
	Direction	in
Code	*PORT_CLOCK = BIT_CLOCK: 0) ;	= (*PORT_CLOCK & ~ BIT_CLOCK) (value ?
Documentation	Send the Clock sign	nal to the motor driver (8904).

<u>private reset()</u>		
Code *	PORT_RESET =	(*PORT_RESET & ~BIT_RESET) 0; (*PORT_RESET & ~BIT_RESET) BIT_RESET [.]

Documentation	Controls the reset signal on the motor driver (8904).
Documentation	Controls the reset signal on the motor driver (8904).

<pre>public motor_running_time_decrease() : void</pre>		
Code	<pre>if(motor_running_time>0) { motor_running_time; } else if (motor_running_time==0) { stopRotation(); }</pre>	
Documentation	Decreases the motor_running_time by 1 ms.	

public EnableWheel() : void		
Code	$motor_enable = 0x1;$	
Documentation	Enables reaction wheel. Any further action of reaction wheel will be carried on.	

public DisableWheel() : void		
Code	$motor_enable = 0x0;$	
Documentation	Disables reaction wheel. Any further action of reaction wheel will be disregarded.	

5.2.2 Gyroscope

Controls the motor by setting the desired speed to the motor. It generates and sends necessary signals to the motor driver 8904 which in turns control the motor.

Gyroscope Class consists of :

- 1) Attributes
- 2) Operations

Gyroscope ->Attributes

private Temp Min : short = -40		
Documentation	This is the temperature corresponding to which the ADC10 temperature output is 0 (minimum). This is the minimum temperature in °C which the Gyroscope (ADXRS401) can read.	
private Temp Ma	<u>ax : short = 85</u>	
Documentation	This is the temperature corresponding to which the ADC10 temperature output is 1023/4095(maximum). The maximum temperature in °C which the Gyroscope (ADXRS401) can read.	
private Speed M	<u>in : short = -75</u>	
Documentation	This is the spin corresponding to which the ADC10 spin output is 0 (minimum). This is the minimum spin in °/s which the Gyroscope (ADXRS401) can read.	
private Speed Ma	<u>ax : short = 75</u>	
Documentation	This is the spin corresponding to which the ADC10 spin output is 1023/4095 (maximum). This is the maximumspin in °/s which the Gyroscope (ADXRS401) can read.	
private GYROSC	<u>COPE Z: int = INCH 2</u>	
Documentation	the ADC channel on which gyroscope spin output is connected.	

private GYROSCOPE T : int = INCH 3		
Documentation	the ADC channel on which gyroscope temperature output is connected.	
<u>public ADC max value : short = 1023</u>		
Documentation	Maximum output of ADC. 1023 for 10-bit ADC. 4095 for 12-bit ADC.	

Gyroscope ->**Operations**

public init_GYR	D ()
Code	<pre>config.gyroscope_gain_s = (unsigned short) (1000.0 * (Speed_Max - Speed_Min) / ADC_max_value); config.gyroscope_offset_s = (unsigned short) (1000.0 * (0 - Speed_Min) / config.gyroscope_gain_s); config.gyroscope_gain_T = (unsigned short) (1000.0 * (Temp_Max - Temp_Min) / ADC_max_value); // ADC_max_value = 1023 for 10-bit ADC config.gyroscope_offset_T = (unsigned short) (1000.0 * (0 - Temp_Min) / config.gyroscope_gain_T); //gyroscope_offset_T is 327 which is the ADC o/p for 0°C //adc ADC():</pre>
Documentation	It initializesgyroscope_gain_s,gyroscope_gain_T,gyroscope_offset_s and gyroscope_offset_T to the required value. These values should be adjusted with correct ones obtained by calibration process.

<pre>public read_spin() : t_sensor</pre>		
Code	t_sensor val;	
	t_sensor spin;	
	adc.select(GYROSCOPE_Z,4);	
	adc.start();	
	while (adc.isReady()==false); val=adc read():	
	spin = (t_sensor)((val - config.gyroscope_offset_s) * (long)	
	config.gyroscope_gain_s); return spin:	
Documentation	It measures and returns the rotational speed around z axis with a signed 16- bits integer; unit is 0.0001 rad/s.	
	This operation uses the ADC(from Common) that returns a digital value.	
	This value is software adjusted, offsetted by <i>gyroscope_offset_s and then</i> multiplied by <i>gyroscope_gain_s</i> and to obtain correct value for spin measure.	
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public read_temp() : t_sensor		
Code	t_sensor val;	
	t_sensor temperature,	
	adc.select(GYROSCOPE_Z,4);	
	adc.start(); while (adc.isReady()==false);	
	val=adc.read();	
	temperature = (t_sensor)(((val - config.gyroscope_offset_T) * (long) config.gyroscope_gain_T) / 10) ; return temperature;	
Documentation	It measures and returns the temperature with a signed 16-bits integer; unit is 0.01°C.	
	This operation calls ADC that returns a value between 0 and ADC_max_value (1023 or 4095).	
	This value is software adjusted, divided by $gyroscope_gain_T$ after being offset by $gyroscope_offset_T$ which is the ADC output for temperature = 0°C.	

CONCLUSION

All the modules for 'Reaction Wheel Actuator' and 'Gyroscopic Sensor' have been successfully coded using UML in C++. The verification has been done by downloading the code onto the circuit with the microcontroller TI MSP430F2274. All the modules necessary for the communication with the central processor have been written and tested. The modelling for the system has been done in UML through Visual Paradigm and the compilation, simulation and assembly code generation along with downloading onto the chip has been done by IAR Embedded Workbench for MSP430.

Personal Comments on Internship and Stay in Torino

I was fortunate enough to get this offer to come to Politecnico di Torino, Italy for a summer internship. This internship has increased my technical, interpersonal and communication skills significantly through the enriching exposure of the technology, culture and society. The technology and infrastructure present here is as expected 'great'. Working on the development of a satellite is quite interesting. The thought that 'you are among the team which is developing a satellite, which is going to be launched soon' is itself great and motivating.

I learnt many new software, tools, techniques and analysed various circuits. Further I developed modules and programmed the microcontroller to control the spin of the satellite. I gained a much better understanding of the satellites and how they are developed. The meetings which I attended(even though quite a few) gave me an idea of professional discussions and how multi-disciplinary projects are coordinated and handled. All these things will surely help me in my future.

My stay at Torino, Italy has been very comfortable and pleasant. I have been living at Borsellino Residence near Politecnico. The residence is quite nice providing all the necessary facilities including well furnished room, Internet, Game Room, TV Room, 24 hrs Reception, Laundry etc.

Also the transport system of the city is nice and convenient to use. The railway station Torino Porta Nuova have trains to all major places in Italy and in Europe. Due to it, very few of my weekends have been in Torino, mostly I went for some visit to some good place on the weekends. I liked this country a lot(as I did a lot of travelling) and even the people are quite friendly.

I will surely like to come here again in future if I get a chance (for Masters or another internship).

Actually the way I got this internship was little amazing and surprising.

I was in my final year of my graduation B.E. (Hons) Electrical & Electronics from BITS Pilani, India. I applied to the professor long back for a 6-months internship from January-June 2009, but unfortunately didn't get through. No regrets, I did the internship in a company in New Delhi, India near my home.

Fortunately I have also got a Job in a reputed multinational company in Bangalore (CISCO Systems Inc., Bangalore, <u>www.cisco.com</u>). I was just waiting for the starting of the job and came to know that it will be probably from 1st July, 2009 while my current internship would be ending by 14th June, 2009.

I was just waiting for the internship to end and job to begin. Suddenly one morning I checked my mail and was surprised to see an email from the Professor Leonardo which said that he has got extra funding and wants to call me for an internship and asked if I am still available?? ⁽ⁱ⁾

The first thought that came to my mind is... 'ohhh no, how is this possible'. Alas, I can't go, because my job joining is just after the current internship and I don't have any time in between. I tried to postpone my joining, but that wasn't an option I had. Still I told the professor that I am interested and asked for the details. But unluckily didn't get any reply for many days. I simply interpreted that I can't go for this and should better forget it.

But suddenly few days after I came to know(unofficially) that the job joining is getting postponed (because of recession, or company policy or whatever) and simultaneously got a confirmed reply from

the professor about the internship. There I was: everything getting rearranged automatically and paving way for me to go Italy. Few more formalities and discussion over the dates of internship and 'All set'.

Finishing previous internship in Delhi on 13th June, fly to Italy on 14th June, back to home from Italy on 13th August and fly to Bangalore for job on 16th August , everything well, internship fits exactly in the time which otherwise would have been vacations.

A trip to Italy (and even to Switzerland, Spain, Belgium, France....) and that too officially, along with a nice experience with the latest technology in the field and chance to work on the Satellites, what else can be better than this??....